

Effects of Hall Current on Unsteady MHD Couette Flow of Class-II in a Rotating System

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ABSTRACT

Unsteady hydromagnetic Couette flow of class-II of a viscous incompressible electrically conducting fluid in a rotating system with Hall effects in the presence of a uniform transverse magnetic field is studied. Both the fluid and plates of the channel are assumed to be at rest when time $t' \leq 0$ and fluid flow within the channel is induced due to non-torsional oscillations of the upper plate in its own plane with a velocity $U(t')$ about a non-zero uniform velocity U_0 at time $t' > 0$. Exact solution of the governing equations is obtained by Laplace transform technique. Asymptotic behavior of the solution is analyzed for small and large values of rotation parameter $K²$ and magnetic parameter $M²$ when time $t>>1$. The numerical values of the fluid velocity are depicted graphically whereas that of shear stress at the plates are presented in tabular form for various values of Hall current parameter m , rotation parameter K^2 , magnetic parameter M^2 and frequency parameter ω .

Keywords: MHD Couette flow of class-II, Rotation, Oscillations, Stokes-Ekman boundary layers, Modified Ekman boundary layers, Hartmann boundary layer.

NOMENCLATURE

1. INTRODUCTION

Investigation of unsteady hydromagnetic flow of a viscous, incompressible and electrically conducting fluid assumes significance because transient nature of fluid flow may be expected at the start-up time of so many MHD devices viz. MHD energy generators, MHD pumps, induced type pumps used in nuclear reactors, MHD accelerators, MHD flow meters etc.. Keeping in view this fact unsteady MHD Couette flow ional primary and secondary pectively

- oordinates
-
- mductivity
- oefficient of viscosity
- arameter
- zular velocity
- lth variable
- , τ_x , τ_y nd secondary shear stress respectively

of a viscous, incompressible and electrically conducting fluid is investigated by a number of researchers. Mentioned may be made of the research studies of Tao (1960), Katagiri (1962), Muhuri (1963), Soundalgekar (1967), Mishra and Muduli (1980), Singh and Kumar (1983) and Seth *et al*. (2011a). Unsteady hydromagnetic Couette flow in a rotating system finds widespread applications in geophysics, planetary sciences and also in many areas of industrial engineering. In such types of

flow Coriolis and magnetic forces play an important role in determining flow features of the problem. It may be noted that Coriolis force is much stronger than the inertial and viscous forces and it is comparable in magnitude with magnetic force. Keeping in view importance of such fluid flow problems Seth *et al*. (1982, 1988, 2009, 2010a, 2010b, 2011b, 2012), Chandran *et al*. (1993), Singh *et al*. (1994), Singh (2000), Ghosh and Pop (2004), Hayat *et al*. (2004a, 2004b, 2004c), Das *et al*. (2009), Guria *et al*. (2009) and Beg *et al*. (2011) investigated unsteady MHD Couette flow of a viscous, incompressible and electrically conducting fluid in a rotating system considering different aspects of the problem. On the basis of the above mentioned research studies on MHD Couette flow we are of opinion that MHD Couette flow may be induced in two ways and it can be put into two classes, namely, (i) MHD Couette flow of class-I and (ii) MHD Couette flow of class-II. The fluid flow which is induced due to the movement of a plate, when fluid is bounded by a stationary plate placed at a finite distance from the moving plate, may be recognized as MHD Couette flow of class-I. This fluid flow is similar to the fluid flow induced by a moving plate when free stream is stationary. The fluid flow past a stationary plate which is induced due to movement of a plate placed at a finite distance from the stationary plate may identified as MHD Couette flow of class-II. This fluid flow is similar to the flow past a stationary plate due to moving free stream. Research studies carried out by Seth *et al*. (1982, 1988, 2010a, 2010b, 2011b, 2012), Chandran *et al*. (1993), Singh *et al*. (1994), Ghosh (2002), Ghosh and Pop (2004), Guria *et al*. (2008, 2009), Das *et al*. (2009) and Beg *et al*. (2011) belong to MHD Couette flow of class-I whereas research investigations of Singh (2000), Hayat *et al*. (2004a, 2004b, 2004c), Seth *et al*. (2009, 2011c), Seth and Singh (2011) belong to MHD Couette flow of class-II.

It is well known that in an ionized fluid, where density is very low and/or the magnetic field is strong, the effects of Hall current become significant as mentioned by Cowling (1957) because Hall current induces secondary flow in the flow-field. Taking into account this fact Ghosh and Pop (2004) and Seth *et al*. (2012) studied Hall effects on unsteady MHD Couette flow in a rotating environment which belongs to MHD Couette flow of class-I whereas Hayat *et al*. (2004c) and Seth *et al*. (2009) investigated effects of Hall current on oscillatory MHD Couette flow of class-II in a rotating system considering different aspects of the problem.

Present investigation deals with the study of unsteady MHD Couette flow of class-II of a viscous, incompressible and electrically conducting fluid in a rotating system with Hall effects in the presence of a uniform transverse magnetic field applied parallel to the axis of rotation. Both the fluid and plates of the channel are assumed to be at rest when time $t' \leq 0$ and fluid flow within the channel is induced due to non-torsional oscillations of the upper plate in its own plane with a velocity $U(t')$ about a non-zero uniform velocity U_0

at time $t' > 0$. Such study assumes importance because both Hall current and rotation induce secondary flow (i.e. cross flow) in the flow-field.

2. MATHEMATICAL FORMULATION AND SOLUTION

Consider flow of a viscous, incompressible and electrically conducting fluid between two infinite parallel plates $z = 0$ and $z = L$ in the presence of a uniform transverse magnetic field B_0 which is applied parallel to *z* -axis. The fluid and channel rotate in counter clockwise direction with a uniform angular velocity Ω about *z*-axis. Both the fluid and plates of the channel are assumed to be at rest at time $t' \leq 0$. Upper plate $z = L$ starts executing non-torsional oscillations in its own plane with a velocity $U(t')$ about a non-zero uniform mean velocity U_0 in *x*direction at time $t' > 0$ and the lower plate $z = 0$, which coincides with *xy*-plane, is kept fixed. Geometry of the problem is presented in Fig.1. It is assumed that the induced magnetic field produced by fluid motion is neglected in comparison to applied one. This is justified because magnetic Reynolds number is very small for metallic liquid and partially ionized fluids (Cramer and Pai, 1973). Also no applied or polarization voltage exists i.e. electric field $\vec{E} \equiv 0$. This corresponds to the case when no energy is added or extracted from the fluid by electrical means (Meyer, 1958). Since plates are of infinite extent in x and y -directions, all

physical quantities except pressure depend on *z* and *t* only.

Fig. 1. Geometry of the problem

In view of the assumptions made above the governing equations for the fluid flow problem in a rotating system with Hall effects are given by

$$
\frac{\partial u}{\partial t'} - 2\Omega v
$$

= $-\frac{1}{\rho} \frac{\partial p^*}{\partial x} + v \frac{\partial^2 u}{\partial z^2} - \frac{\sigma B_0^2}{\rho (1 + m^2)} (u - mv)$ (1)

$$
\frac{\partial v}{\partial t'} + 2\Omega u
$$

= $-\frac{1}{\rho} \frac{\partial p^*}{\partial y} + v \frac{\partial^2 v}{\partial z^2} - \frac{\sigma B_0^2}{\rho (1 + m^2)} (v + mu)$ (2)

$$
0 = -\frac{1}{\rho} \frac{\partial p^*}{\partial z}
$$
 (3)

where *u*, *v*, *v*, σ , ρ , $m = \omega_e \tau_e$, ω_e , τ_e and p^* are, respectively, fluid velocity in x-direction, fluid velocity in y-direction, kinematic coefficient of viscosity, electrical conductivity of the fluid, fluid density, Hall current parameter, cyclotron frequency, electron collision time and modified pressure including centrifugal force.

The initial and boundary conditions are given by

$$
u = v = 0 \text{ for } 0 \le z \le L \quad \text{and } t' \le 0 \tag{4}
$$

$$
u = v = 0
$$
 at $z = 0$ for $t' > 0$ (5)

$$
u = V = 0
$$
 or $u = 0$ for $t > 0$ (3)
 $u = U(t'), v = 0$ at $z = L$ for $t' > 0$ (6)

Equation (3) shows that modified pressure p^* is uniform along *z* -axis i.e. axis of rotation. Taking into the consideration of research studies made on MHD Couette flow till now, we are of the opinion that MHD Couette flow may be classified in two forms, namely, (i) MHD Couette flow of class-I and (ii) MHD Couette flow of class-II. The fluid flow induced due to movement of a plate, when fluid is bounded by a stationary plate placed at a finite distance from the moving plate, may be identified as MHD Couette flow of class-I. This fluid flow is similar to the flow induced due to movement of a plate when free stream is stationary. The fluid flow past a stationary plate, which is induced due to movement of a plate placed at a finite distance from the stationary plate, may be regarded as MHD Couette flow of class-II. This fluid flow is similar to the flow past a stationary plate due to moving free stream. For unsteady MHD Couette flow of class-I the

pressure gradient terms
$$
-\frac{1}{\rho} \frac{\partial p^*}{\partial x}
$$
 and $-\frac{1}{\rho} \frac{\partial p^*}{\partial y}$, which

are present in Eqs. (1) and (2) respectively, are not considered by Seth *et al* (1982, 1988, 2010a, 2010b, 2011b, 2012), Chandran *et al*. (1993), Singh *et al*. (1994), Ghosh (2002), Ghosh and Pop (2004), Guria *et al*. (2008, 2009), Das *et al*. (2009) and Beg *et al*. (2011). This assumption is justified and it is clearly evident from Eq. (5). For unsteady MHD Couette flow of class-II values of pressure gradient terms in Eqs. (1) and (2) are obtained with the help of boundary conditions, Eq. (6) which are given below

$$
-\frac{1}{\rho} \frac{\partial p^*}{\partial x} = \frac{\partial U}{\partial t'} + \frac{\sigma B_0^2 U}{\rho (1 + m^2)},
$$

$$
-\frac{1}{\rho} \frac{\partial p^*}{\partial y} = 2U \Omega + \frac{\sigma m B_0^2 U}{\rho (1 + m^2)}
$$
(7)

Making use of Eq. (7) , the Eqs. (1) and (2) reduce to

$$
\frac{\partial u}{\partial t'} - 2\Omega v
$$
\n
$$
= \frac{\partial U}{\partial t'} + v \frac{\partial^2 u}{\partial z^2} - \frac{\sigma B_0^2}{\rho (1 + m^2)} \Big[(u - U) - mv \Big]
$$
\n(8)

$$
\frac{\partial v}{\partial t'} + 2\Omega(u - U) \n= v \frac{\partial^2 v}{\partial z^2} - \frac{\sigma B_0^2}{\rho(1 + m^2)} \Big[v + m(u - U) \Big]
$$
\n(9)

Equations (8) and (9), in non-dimensional form, become

$$
\frac{\partial u_1}{\partial t} - 2K^2 v_1
$$
\n
$$
= \frac{\partial F}{\partial t} + \frac{\partial^2 u_1}{\partial \eta^2} - \frac{M^2}{1 + m^2} \left[(u_1 - F) - mv_1 \right]
$$
\n(10)

$$
\frac{\partial v_1}{\partial t} + 2K^2(u_1 - F)
$$

=
$$
\frac{\partial^2 v_1}{\partial \eta^2} - \frac{M^2}{1 + m^2} [v_1 + m(u_1 - F)]
$$
 (11)

where $\eta = z / L$, $u_1 = u / U_0$, $v_1 = v / U_0$, $U = U_0 F(t)$, $M^2 = L^2 B_0^2 (\sigma / \rho v)$ is magnetic parameter which is square of Hartmann number and $K^2 = \Omega L^2 / v$ is rotation parameter which is reciprocal of Ekman number.

The initial and boundary conditions, Eq. (4) to Eq. (6), in non-dimensional form, become

$$
u_1 = v_1 = 0 \qquad \text{for } 0 \le \eta \le 1 \quad \text{and } t \le 0 \qquad (12)
$$

$$
u_1 = v_1 = 0
$$
 at $\eta = 0$ for $t > 0$ (13)

$$
u_1 = F(t), v_1 = 0
$$
 at $\eta = 1$ for $t > 0$ (14)

Combining Eq. (10) with Eq. (11) , we obtain

$$
\frac{\partial q}{\partial t} + i \left[\frac{mM^2}{1 + m^2} + 2K^2 \right] (q - F)
$$

= $\frac{\partial F}{\partial t} + \frac{\partial^2 q}{\partial \eta^2} - \frac{M^2}{1 + m^2} (q - F)$ (15)

where $q = u_1 + iv_1$.

The initial and boundary conditions, Eq. (12) to Eq. (14), in compact form, become

 $q = 0$ for $0 \le \eta \le 1$ and $t \le 0$ (16)

$$
q = 0 \qquad \text{at } \eta = 0 \qquad \text{for } t > 0 \qquad (17)
$$

$$
q = F(t) \qquad \text{at } \eta = 1 \qquad \text{for } t > 0 \qquad (18)
$$

Since fluid flow is of oscillatory nature we may assume $F(t)$ in the following form

$$
F(t) = 1 + (ae^{i\omega t} + be^{-i\omega t})
$$
\n(19)

where a and b are complex constants and b is complex conjugate of *a*. $\omega = \omega' L^2 / \omega$ is frequency parameter where ω' is frequency of oscillations.

Using Laplace transform, Eq. (15) subject to the initial condition, Eq. (16) reduces to

$$
\frac{d^2\overline{q}}{d\eta^2} - \left[p + \frac{M^2}{1+m^2} + i \left(\frac{mM^2}{1+m^2} + 2K^2 \right) \right] \overline{q}
$$

=
$$
- \left[p + \frac{M^2}{1+m^2} + i \left(\frac{mM^2}{1+m^2} + 2K^2 \right) \right] \overline{F}
$$
(20)

where $\overline{q} = \int_0^\infty e^{-pt} q(\eta, t) dt$, $\overline{F} = \int_0^\infty e^{-pt} F(t) dt$ $=\int_0^\infty e^{-pt}q(\eta,t)dt, \ \overline{F}=\int_0^\infty e^{-pt}F(t)dt$ and *p* being Laplace transform parameter.

Boundary conditions, Eq. (17) and Eq. (18) with use of Eq. (19) after taking Laplace transform become

$$
\overline{q} = 0 \qquad \text{at } \eta = 0 \tag{21}
$$

$$
\overline{q} = \overline{F}(p) = \frac{1}{p} + \frac{a}{p - i\omega} + \frac{b}{p + i\omega} \quad \text{at } \eta = 1 \quad (22)
$$

Solution of Eq. (20) subject to the boundary conditions, Eq. (21) and Eq. (22) is given by

$$
\overline{q} = \left(\frac{1}{p} + \frac{a}{p - i\omega} + \frac{b}{p + i\omega}\right)
$$

$$
\times \left(1 - \frac{\sinh \lambda (1 - \eta)}{\sinh \lambda}\right)
$$
 (23)

where
$$
\lambda = \left[p + M^2 + iN^2 \right]^{1/2}
$$
 and $N^2 = \frac{mM^2}{1 + m^2} + 2K^2$.

Taking inverse Laplace transform of Eq. (23), we obtain

$$
q(\eta, t)
$$
\n
$$
= \left[1 - \frac{\sinh(\alpha_1 + i \beta_1)(1 - \eta)}{\sinh(\alpha_1 + i \beta_1)}\right]
$$
\n
$$
+ \left[1 - \frac{\sinh(\alpha_2 + i \beta_2)(1 - \eta)}{\sinh(\alpha_2 + i \beta_2)}\right] a e^{i \omega t}
$$
\n
$$
+ \left[1 - \frac{\sinh(\alpha_3 \pm i \beta_3)(1 - \eta)}{\sinh(\alpha_3 \pm i \beta_3)}\right] b e^{-i \omega t}
$$
\n
$$
- \sum_{n=1}^{\infty} 2n \pi \sin n \pi \eta \left(\frac{1}{s_1} + \frac{a}{s_1 - i \omega} + \frac{b}{s_1 + i \omega}\right) e^{s_1 t}
$$

for $\omega \neq N^2$

where

$$
\alpha_1 = \frac{1}{\sqrt{2}} \left[\left\{ \frac{M^4}{(1+m^2)^2} + N^4 \right\}^{1/2} + \frac{M^2}{1+m^2} \right]^{1/2},\tag{25}
$$

$$
\beta_1 = \frac{1}{\sqrt{2}} \left[\left\{ \frac{M^4}{(1+m^2)^2} + N^4 \right\}^{1/2} - \frac{M^2}{1+m^2} \right]^{1/2},
$$

\n
$$
\alpha_2 = \frac{1}{\sqrt{2}} \left[\left\{ \frac{M^4}{(1+m^2)^2} + \left(\omega + N^2\right)^2 \right\}^{1/2} + \frac{M^2}{1+m^2} \right]^{1/2},
$$

\n
$$
\beta_2 = \frac{1}{\sqrt{2}} \left[\left\{ \frac{M^4}{(1+m^2)^2} + \left(\omega + N^2\right)^2 \right\}^{1/2} - \frac{M^2}{1+m^2} \right]^{1/2},
$$

\n
$$
\alpha_3 = \frac{1}{\sqrt{2}} \left[\left\{ \frac{M^4}{(1+m^2)^2} + \left(\omega - N^2\right)^2 \right\}^{1/2} + \frac{M^2}{1+m^2} \right]^{1/2},
$$

\n
$$
\beta_3 = \frac{1}{\sqrt{2}} \left[\left\{ \frac{M^4}{(1+m^2)^2} + \left(\omega - N^2\right)^2 \right\}^{1/2} - \frac{M^2}{1+m^2} \right]^{1/2},
$$

\nand
$$
s_1 = -\left[n^2 \pi^2 + \frac{M^2}{1+m^2} + iN^2 \right]
$$

The upper and lower signs in Eq. (24) are considered for $\omega < N^2$ and $\omega > N^2$ respectively.

When $\omega = N^2$ i.e. when the natural frequency N^2 , which is due to rotation and Hall current, is equal to the impressed frequency ω the inverse Laplace transform of Eq. (23) gives

$$
q(\eta, t)
$$
\n
$$
= \left[1 - \frac{\sinh(\alpha'_1 + i \beta'_1)(1 - \eta)}{\sinh(\alpha'_1 + i \beta'_1)} \right]
$$
\n
$$
+ \left[1 - \frac{\sinh(\alpha'_2 + i \beta'_2)(1 - \eta)}{\sinh(\alpha'_2 + i \beta'_2)} \right] \stackrel{\text{a.e. } i \text{ or } j}{=} \pi
$$
\n
$$
+ \left[1 - \frac{\sinh \alpha'_3(1 - \eta)}{\sinh \alpha'_3} \right] b e^{-i \text{ or } j}
$$
\n
$$
- \sum_{n=1}^{\infty} 2n \pi \sin n \pi \eta \left(\frac{1}{s_1'} + \frac{a}{s_1' - i \omega} + \frac{b}{s_1' + i \omega} \right) e^{s_1' t}
$$
\n(26)

where

$$
\alpha_1' = \frac{1}{\sqrt{2}} \left[\left\{ \frac{M^4}{(1+m^2)^2} + \omega^2 \right\}^{1/2} + \frac{M^2}{1+m^2} \right]^{1/2},
$$

$$
\beta_1' = \frac{1}{\sqrt{2}} \left[\left\{ \frac{M^4}{(1+m^2)^2} + \omega^2 \right\}^{1/2} - \frac{M^2}{1+m^2} \right]^{1/2}, \qquad (27)
$$

$$
\alpha_2' = \frac{1}{\sqrt{2}} \left[\left\{ \frac{M^4}{(1+m^2)^2} + 4\omega^2 \right\}^{1/2} + \frac{M^2}{1+m^2} \right]^{1/2},
$$

$$
\beta_2' = \frac{1}{\sqrt{2}} \left[\left\{ \frac{M^4}{(1+m^2)^2} + 4\omega^2 \right\}^{1/2} - \frac{M^2}{1+m^2} \right]^{1/2},
$$

\n
$$
\alpha_3' = \frac{M}{\sqrt{(1+m^2)}},
$$

\nand
$$
s_1' = -\left[n^2 \pi^2 + \frac{M^2}{1+m^2} + i\omega \right].
$$
 (27)

Equations $(24)-(27)$ represent the solution for fluid velocity in general case. The solutions, Eq. (24) and Eq. (26) exhibit a unified representation of initial MHD Couette flow induced due to non-torsional oscillations of the upper plate, final steady state flow and decaying oscillations excited by interaction of magnetic field, Coriolis force, Hall current and initial oscillatory motion. In the absence of magnetic field (i.e. $M^2 = 0$) and Hall current (i.e. $m = 0$) solutions, Eq. (24) and Eq. (26) are in agreement with the solutions obtained by Das *et al*. (2008).

We shall now examine the solutions, Eq. (24) and Eq. (26) for large values of time *t* and for small as well large values of rotation parameter K^2 and magnetic parameter M^2 .

It may be noted that when time t is large i.e. $t>>1$ then ω is small (i.e. $\omega \ll 1$) such that ωt is finite. For large time t and $a = b = \varepsilon / 2$ the solutions, Eq. (24) and Eq. (26) assume the following form

$$
q(\eta, t)
$$
\n
$$
= \left[1 - \frac{\sinh(\alpha_1 + i \beta_1)(1 - \eta)}{\sinh(\alpha_1 + i \beta_1)}\right]
$$
\n
$$
+ \frac{\varepsilon}{2} \left[\left\{1 - \frac{\sinh(\alpha_2 + i \beta_2)(1 - \eta)}{\sinh(\alpha_2 + i \beta_2)}\right\} e^{i \omega t} + \left\{1 - \frac{\sinh(\alpha_3 \pm i \beta_3)(1 - \eta)}{\sinh(\alpha_3 \pm i \beta_3)}\right\} e^{-i \omega t}\right]
$$
\n(28)\n
$$
\text{for } \omega \neq N^2
$$

$$
q(\eta, t)
$$
\n
$$
= \left[1 - \frac{\sinh(\alpha_1' + i \beta_1')(1 - \eta)}{\sinh(\alpha_1' + i \beta_1')} \right]
$$
\n
$$
+ \frac{\varepsilon}{2} \left[\left\{1 - \frac{\sinh(\alpha_2' + i \beta_2')(1 - \eta)}{\sinh(\alpha_2' + i \beta_2')} \right\} e^{i \alpha t} \right]
$$
\n
$$
+ \left\{1 - \frac{\sinh \alpha_3'(1 - \eta)}{\sinh \alpha_3'} \right\} e^{-i \alpha t} \right]
$$
\nfor $\omega = N^2$ (29)

The solutions, Eq. (28) and Eq. (29) represent the solution for fully developed oscillatory hydromagnetic Couette flow in a rotating system. In the absence of the Hall current (i.e. $m = 0$) solutions, Eq. (28) and Eq. (29) are in agreement with the solutions obtained by Singh (2000).

3. ASYMPTOTIC SOLUTION

We shall now examine the asymptotic behavior of the solution (28) for small as well as large values of K^2 and M^2 when $\varepsilon = 1$ to gain some physical insight into the flow pattern.

Case I: $K^2 \ll 1$ and $M^2 \ll 1$

Since M^2 , K^2 and ω are very small, neglecting square and higher powers of M^2 , K^2 and ω in Eq. (28), primary velocity u_1 and secondary velocity v_1 assume the following form

$$
u_1 = 1 + \cos \omega t - (1 - \eta) \left[1 + \cos \omega t + \frac{1}{6} (\eta^2 - 2\eta) + \frac{1}{6} (\eta^2 - 2\eta) \right]
$$
\n
$$
\times \left\{ \frac{M^2}{1 + m^2} (1 + \cos \omega t) - \omega \sin \omega t \right\} \right] + \cdots
$$
\n(30)

$$
v_1 = -\frac{N^2}{6}(1-\eta)(\eta^2 - 2\eta)(1 + \cos \omega t) + \cdots
$$
 (31)

It is evident from the Eq. (30) and Eq. (31) that for a slowly rotating system with small frequency of oscillations when the conductivity of the fluid is low and/or the applied magnetic field is weak, primary velocity u_1 is independent of rotation while the secondary velocity v_1 is affected by magnetic field, Hall current and rotation. The fluid flow in both the directions has considerable effects of oscillations. In the absence of Hall current secondary velocity v_1 is unaffected by magnetic field.

Case II:
$$
K^2 >> 1
$$
 and $M^2 \sim O(1)$

When K^2 is very large and M^2 is small order of magnitude fluid flow becomes boundary layer type. For the boundary layer flow near the lower plate $\eta = 0$, primary velocity u_1 and secondary velocity v_1 in Eq. (28) assume the following form

$$
u_1 = 1 + \cos \omega t - e^{-\alpha_4 \eta} \cos \beta_4 \eta
$$

$$
- \frac{1}{2} \Big[e^{-\alpha_5 \eta} \cos(\omega t - \beta_5 \eta) + e^{-\alpha_6 \eta} \cos(\omega t + \beta_6 \eta) \Big]
$$
(32)

$$
v_1 = e^{-\alpha_4 \eta} \sin \beta_4 \eta - \frac{1}{2} \left[e^{-\alpha_5 \eta} \sin(\omega t - \beta_5 \eta) - e^{-\alpha_6 \eta} \sin(\omega t \pm \beta_6 \eta) \right]
$$
(33)

where

$$
\alpha_4 = K \left[1 + \frac{(m+1)M^2}{4(1+m^2)K^2} \right],
$$
\n(34)

$$
\beta_4 = K \left[1 + \frac{(m-1)M^2}{4(1+m^2)K^2} \right],
$$
\n
$$
\alpha_5 = K \left[1 + \frac{\omega}{4K^2} + \frac{(m+1)M^2}{4(1+m^2)K^2} \right],
$$
\n
$$
\beta_5 = K \left[1 + \frac{\omega}{4K^2} + \frac{(m-1)M^2}{4(1+m^2)K^2} \right],
$$
\n
$$
\alpha_6 = K \left[1 - \frac{\omega}{4K^2} + \frac{(m+1)M^2}{4(1+m^2)K^2} \right],
$$
\n
$$
\beta_6 = K \left[1 - \frac{\omega}{4K^2} + \frac{(m-1)M^2}{4(1+m^2)K^2} \right].
$$
\n(34)

It is revealed from the expressions Eq. (32) and Eq. (33) that the solution is in quasi-steady state. Steady state flow is confined within a boundary layer of thickness $O(1/\alpha_4)$ which may be identified as modified Ekman boundary layer and can be viewed as classical Ekaman boundary layer modified by Hall current and magnetic field. It is noticed from the expressions Eq. (32) and Eq. (33) that the unsteady flow has three modes of oscillations. The first mode corresponds to the pure oscillations of frequency ω due to non-torsional oscillations of upper plate of the channel which persist in the entire fluid region. The other two modes of oscillations correspond to modified Stokes flow and are confined within boundary layers of thickness $O(1/\alpha_5)$ and $O(1/\alpha_6)$. These boundary layers may be recognized as modified Stokes-Ekaman boundary layers and may be viewed as classical Ekman bounadry layers modified by magnetic field, oscillations and Hall current. It may be noted from Eq. (34) that α_4 -layer is thicker than α_5 -layer whereas α_6 -layer is thicker than α_4 -layer. The thickness of α_4 -layer decreases with increase in either K^2 or M^2 whereas it increases with increase in m . The thickness of α_5 -layer decreases with increase in either K^2 or M^2 or ω whereas it increases with increase in m . The thickness of α_6 layer decreases with increase in either K^2 or M^2 whereas it increases with increase in either ω or m . Similar type of boundary layers arise in the neighborhood of the upper plate. It is evident from Eq. (32) and Eq. (33) that unsteady flow is divided into two parts. One part oscillates with amplitude $\frac{1}{2}e^{-\alpha_5}$ $\frac{1}{2}e^{-\alpha_5 \eta}$ and the other one with $\frac{1}{2}e^{-\alpha_6}$ $\frac{1}{2}e^{-\alpha_6\eta}$. The unsteady flow corresponding to the former part oscillates with phase lag of $\beta_5 \eta$ whereas the unsteady flow corresponding to latter part oscillates with a phase lead of $\beta_6\eta$.

The exponential terms in Eq. (32) and Eq. (33) damped out quickly as η increases. When $\eta \ge 1/\alpha_6$ i.e. outside the boundary layer region, Eq. (32) and Eq. (33) reduce to

$$
u_1 \approx 1 + \cos \omega t, \, v_1 \approx 0 \tag{35}
$$

It is evident from the expression Eq. (35) that, in a certain core given by $\eta \geq 1/\alpha_6$ i.e. outside the boundary layer region, the fluid has velocity in primary flow direction which oscillates with the same frequency ω as that of the upper plate when $t > 0$.

Case III: M^2 >>1 and $K^2 \sim O(1)$

This case also corresponds to the boundary layer type flow. For the boundary layer flow near the lower plate $\eta = 0$, we obtain fluid velocity from Eq. (28) as

$$
u_1 = 1 + \cos \omega t - e^{-\alpha_i \eta} \left[\cos \beta_7 \eta + \frac{1}{2} \{ \cos(\omega t - \beta_8 \eta) + \cos(\omega t \pm \beta_9 \eta) \} \right]
$$
(36)

$$
v_1 = e^{-\alpha_i \eta} \left[\sin \beta_7 \eta \right]
$$

$$
1 - e^{-\frac{1}{2}\left\{\sin(\omega t - \beta_8 \eta) - \sin(\omega t \pm \beta_9 \eta)\right\}} \Bigg]
$$
(37)

where

$$
\alpha_{i} = M / \sqrt{(1 + m^{2})} \quad \text{for } i = 7, 8, 9,
$$
\n
$$
\beta_{7} = \frac{N^{2}}{2M / \sqrt{(1 + m^{2})}},
$$
\n
$$
\beta_{8} = \frac{(\omega + N^{2})}{2M / \sqrt{(1 + m^{2})}},
$$
\n
$$
\beta_{9} = \frac{(\omega - N^{2})}{2M / \sqrt{(1 + m^{2})}}.
$$
\n(38)

The expressions Eq. (36) and Eq. (37) show that the fluid flow is in quasi-steady state. Steady state flow is confined within a modified Hartmann boundary layer of thickness $O\left(1/\sqrt{M^2/(1+m^2)}\right)$ which decreases with increase in magnetic parameter $M²$ and increases with

increase in Hall current parameter *m* . Unsteady state flow has three modes of oscillations. The first mode corresponds to pure oscillations with frequency ω due to non-torsional oscillations of the upper plate of the channel when $t > 0$ and fills the entire fluid region. The other two modes of oscillations correspond to the modified Stokes flow and are confined within a thin modified Hartmann boundary layer of thickness $O\left(1/\sqrt{M^2/(1+m^2)}\right)$. Similar type of boundary layer appears adjacent to the upper plate of the channel. It is

interesting to note that unsteady flow is divided into two parts and both the parts oscillate with amplitude $1 - \left(M / \sqrt{1 + m^2}\right)$ 2 $e^{-\left(M/\sqrt{1+m^2}\right)\eta}$. The unsteady flow corresponding to

first part oscillates with phase lag $\beta_8 \eta$ when $\omega \neq N^2$. However, unsteady flow corresponding to second part oscillates with phase lead of $\beta_9 \eta$ when $\omega < N^2$ and this part oscillates with phase lag of $\beta_9 \eta$ when $\omega > N^2$. Outside the boundary layer region, the fluid velocity assumes the form

$$
u_1 \approx 1 + \cos \omega t, \ v_1 \approx 0 \tag{39}
$$

It is revealed from Eq. (39) that fluid flows in primary flow direction only and oscillates with the same frequency ω as that of the upper plate when $t > 0$.

4. SHEAR STRESS AT THE PLATES

The Non-dimensional shear stress at the lower and upper plates due to primary and secondary flows, for $\omega \neq N^2$, are given by

$$
ω > N^2
$$
. Outside the boundary layer region, the fluid
velocity assumes the form
\n $u_1 ≈ 1 + cos αt$, $v_1 ≈ 0$ (39)
\nIt is revealed from Eq. (39) that fluid flows in primary
\nflow direction only and oscillates with the same
\nfrequency ω as that of the upper plate when $t > 0$.
\n4. SHEAR STRESS AT THE PLATES
\nThe Non-dimensional shear stress at the lower and
\nupper plates due to primary and secondary flows, for
\n $ω ≠ N^2$, are given by
\n
$$
\tau_x + i\tau_y |_{\eta=0}
$$
\n
$$
= (α_1 + i β_1)coth(α_1 + i β_1)
$$
\n
$$
+ a(α_2 + i β_2)e^{iα} \coth(α_2 + i β_2)
$$
\n
$$
+ b(α_3 ± i β_3)e^{-iα} \coth(α_3 ± i β_3)
$$
\n
$$
- \sum_{n=1}^{\infty} 2n^2 \pi^2 \left(\frac{1}{s_1} + \frac{a}{s_1 - i \omega} + \frac{b}{s_1 + i \omega} \right) e^{s_1 t}
$$
\n
$$
\tau_x + i\tau_y |_{\eta=1}
$$
\n
$$
= (α_1 + i β_1) cosec (α_1 + i β_1)
$$
\n
$$
+ a(α_2 + i β_2)e^{iα} \csc (α_2 + i β_2)
$$
\n
$$
+ b(α_3 ± i β_3)e^{-iα} \csc (α_3 ± i β_3)
$$
\n(A1)
\n
$$
+ a(α_2 + i β_2)e^{iα} \csc (α_3 ± i β_3)
$$
\n(A2)
\n
$$
+ b(α_3 ± i β_3)e^{-iα} \csc (α_3 ± i β_3)
$$
\n(A3)
\n
$$
+ a(α_2 + i β_2)e^{iα} \csc (α_3 ± i β_3)
$$
\n(A4)
\n
$$
+ a(α_2 + i β_2)e^{iα} \csc (α_3 ± i β_3)
$$
\n(A5)
\n
$$
= \sum_{n=1}^{\infty} (-1)^n 2n^2 \pi^2 \left(\frac{1}{s_1} + \frac{a}{s_1 - i \omega} + \frac{b}{s_1 +
$$

$$
\tau_x + i \tau_y |_{\eta=1}
$$

= $(\alpha_1 + i \beta_1) \csc \alpha (\alpha_1 + i \beta_1)$
+ $a(\alpha_2 + i \beta_2) e^{i \alpha t} \csc \alpha (\alpha_2 + i \beta_2)$
+ $b(\alpha_3 \pm i \beta_3) e^{-i \alpha t} \csc \alpha (\alpha_3 \pm i \beta_3)$

$$
-\sum_{n=1}^{\infty} (-1)^n 2n^2 \pi^2 \left(\frac{1}{s_1} + \frac{a}{s_1 - i \omega} + \frac{b}{s_1 + i \omega}\right) e^{s_1 t}
$$
 (41)

Non-dimensional shear stress at the lower and upper plates due to primary and secondary flows, for $\omega = N^2$, are given by

 \int_{-1}^{1} (-1)² $2n^{-}\pi^{-}$ $\left(\frac{-1}{s_1} + \frac{1}{s_1 - i \omega} + \frac{1}{s_1}\right)$

n

$$
\tau_x + i \tau_y |_{\eta=0}
$$

= $(\alpha'_1 + i \beta'_1) \coth(\alpha'_1 + i \beta'_1)$
+ $a(\alpha'_2 + i \beta'_2) e^{i\omega t} \coth(\alpha'_2 + i \beta'_2)$
+ $b\alpha'_3 e^{-i\omega t} \coth\alpha'_3$

$$
-\sum_{n=1}^{\infty} 2n^2 \pi^2 \left(\frac{1}{s'_1} + \frac{a}{s'_1 - i\omega} + \frac{b}{s'_1 + i\omega} e^{s'_1 t}\right)
$$
 (42)

$$
\tau_x + i\tau_y |_{\eta=1}
$$

= $(\alpha'_1 + i\beta'_1)\csc\alpha(\alpha'_1 + i\beta'_1)$
+ $a(\alpha'_2 + i\beta'_2)e^{i\omega t}\csc\alpha(\alpha'_2 + i\beta'_2)$
+ $b\alpha'_3e^{-i\omega t}\csc\alpha'_3$

$$
-\sum_{n=1}^{\infty} \{(-1)^n 2n^2\pi^2
$$

$$
\times \left(\frac{1}{s'_1} + \frac{a}{s'_1 - i\omega} + \frac{b}{s'_1 + i\omega}\right)\ e^{s'_1 t}
$$
 (43)

5. RESULTS AND DISCUSSION

To study the effects of Hall current, rotation, magnetic field and oscillations on the flow-field the numerical values of the fluid velocity, computed from the analytical solution Eq. (24) mentioned in section 2, are displayed graphically versus channel width variable η in Figs. 2 to 9 for various values of Hall current parameter m , rotation parameter K^2 , magnetic parameter M^2 and frequency parameter ω when $\omega t = \pi / 2$ and $\omega \neq N^2$. It is evident from Fig. 2 and Fig. 3 that primary velocity u_1 decreases whereas secondary velocity v_1 increases on increasing m which implies that Hall current tends to retard primary flow whereas it has reverse effect on secondary flow.

It is revealed from Fig. 4 and Fig. 5 that primary velocity u_1 and secondary velocity v_1 increase on increasing K^2 which implies that rotation tends to accelerate both the primary and secondary flows.

Fig. 4. Velocity profiles when $m = 0.5$, $M^2 = 10$ and $\omega = 5 \left(\omega < N^2 \right)$

It is noticed from Fig. 6 and Fig. 7 that primary velocity u_1 increases whereas secondary velocity v_1 decreases on increasing M^2 which implies that magnetic field tends to accelerate primary flow whereas it has reverse effect on secondary flow.

Fig. 6. Velocity profiles when $m = 0.5$, $K^2 = 2$ and $\omega = 5 \left(\omega \lt N^2 \right)$

Fig. 7. Velocity profiles when $m = 0.5$, $K^2 = 2$ and $\omega = 15 \left(\omega > N^2 \right)$

It is revealed from Fig. 8 and Fig. 9 that primary velocity u_1 decreases on increasing ω when $\omega \neq N^2$ whereas the secondary velocity v_1 increases on increasing ω when $\omega < N^2$ and it decreases on increasing ω when $\omega > N^2$ which implies that oscillations tend to retard primary flow when $\omega \neq N^2$ and secondary flow when $\omega > N^2$ whereas it have reverse effect on secondary flow when $\omega < N^2$.

The numerical values of the primary and secondary shear stress at the lower and upper plates, computed from the analytical expressions Eq. (40) and Eq. (41) mentioned in section 4, are presented in tabular form in Tables 1-8 for various values of m, ω , K^2 and M^2 when $\omega \neq N^2$ by considering $\omega t = \pi / 2$. It is evident from Table 1 that, for $\omega < N^2$, primary shear stress at the lower plate i.e. $\tau_x|_{\eta=0}$ decreases whereas secondary shear stress at the lower plate i.e. $\tau_y|_{\eta=0}$ increases on increasing either m or ω . This implies that Hall current and oscillations tend to reduce primary shear stress at the lower plate whereas it have reverse effect on secondary shear stress at the lower plate when $\omega < N^2$.

Table 1 Shear stress at the lower plate due to primary and secondary flows when $K^2 = 2$

and M^2 – 10($\omega < N^2$)

It is observed from Table 2 that, for $\omega > N^2$, on increasing *m* , $\tau_{\mathbf{x}}|_{\eta=0}$ decreases when ω =13 and 15 and it decreases, attains a minimum and then increases in magnitude when $\omega = 17$ whereas, on increasing ω , $\tau_x|_{\eta=0}$ decreases when $m = 0.5$ and 1.0 and it decreases, attains a minimum and then increases in magnitude when $m = 1.5$. $\tau_y|_{\eta=0}$ increases on increasing *m*. On increasing ω , $\tau_y|_{\eta=0}$ increases when $m=0.5$, it increases, attains a maximum and then decreases when $m = 1.0$ and it decreases when $m = 1.5$. This implies that, for $\omega > N^2$, Hall current tends to enhance secondary shear stress at the lower plate whereas it has reverse effect on the primary shear stress at the lower plate when $\omega \le 15$. Oscillations tend to reduce primary shear stress at the lower plate when $m \le 1.0$ and secondary shear stress at the lower plate behaves in oscillatory manner with respect to oscillations. It may be noted from Table 2 that there exists flow separation at the lower plate in the primary flow direction on increasing either m or ω . It is revealed from Table 3 that, for $\omega < N^2$, primary shear stress at the upper plate i.e. $\tau_{x}|_{\eta=1}$ and secondary shear stress at the upper plate i.e. $\tau_y|_{\eta=1}$ increase on increasing ω .

 $\tau_{\rm x}|_{\eta=1}$ decreases, attains a minimum and then increases on increasing *m*. $\tau_y|_{\eta=1}$ increases on increasing *m*. This implies that, for $\omega < N^2$, Hall current tends to enhance secondary shear stress at the upper plate whereas oscillations have tendency to enhance both the primary and secondary shear stress at the upper plate.

			Table 2 Shear stress at the lower plate due to primary

and secondary flows when $K^2 = 2$ and $M^2 = 10(m > N^2)$

	$\omega \downarrow m \rightarrow$	0.5	1.0	1.5
	13	1.3378	0.8603	0.4007
$\tau_{x} _{\eta=0}$	15	1.1021	0.6027	0.1335
	17	0.8818	0.3674	-0.1031
	13	1.7505	2.2758	2.4749
	15	1.7643	2.2803	2.4585
$\tau_{y} _{\eta=0}$	17	1.7680	2.2701	2.4288

Table 3 Shear stress at the upper plate due to primary and secondary flows when $K^2 = 2$

and $M^2 = 10(\omega < N^2)$

It is noticed from Table 4 that, for $\omega > N^2$, $\tau_x|_{\eta=1}$ and $\tau_y|_{\eta=1}$ increase on increasing *m*. $\tau_x|_{\eta=1}$ increases when $m = 0.5$ and 1.0 and it increases, attains a maximum and then decreases when $m = 1.5$ on increasing ω . $\tau_y|_{\eta=1}$ decreases on increasing ω . This implies that, for $\omega > N^2$, Hall current tends to enhance both the primary and secondary shear stress at the upper plate and oscillations tends to reduce secondary shear stress at the upper plate while it tends to enhance primary shear stress at the upper plate when $m \leq 1.0$. It is observed from Table 5 that, for $\omega < N^2$, $\tau_x |_{\eta=0}$ and $\tau_y |_{\eta=0}$ increase on increasing K^2 . On increasing M^2 , $\tau_x|_{\eta=0}$ decreases when $K^2 = 2$ and it increases when $K^2 = 3$ and 4. On increasing M^2 , $\tau_y|_{\eta=0}$ increases when $K^2 = 2$, it decreases, attains a minimum and then increases when $K^2 = 3$ and it decreases when $K^2 = 4$. This implies that, for $\omega < N^2$, rotation tends to enhance both the primary and secondary shear stress at the lower plate. Magnetic field tends to enhance primary shear stress at the lower plate when $K^2 \geq 3$. Secondary shear stress at the lower plate behaves in oscillatory manner with respect to magnetic field.

Table 4 Shear stress at the upper plate due to primary and secondary flows when $K^2 = 2$

and $M^2 = 10(\omega > N^2)$					
$\omega \downarrow m \rightarrow$		0.5	1.0	1.5	
	13	0.3575	0.3848	0.4826	
$\tau_{x} _{\eta=1}$	15	0.3662	0.4041	0.5021	
	17	0.3665	0.4091	0.5003	
	13	0.4165	0.6368	0.7964	
$-\tau_y _{\eta=1}$	15	0.3977	0.6007	0.7384	
	17	0.3731	0.5559	0.6740	

Table 5 Shear stress at the lower plate due to primary and secondary flows when $m = 0.5$

It is evident from Table 6 that, for $\omega > N^2$, $\tau_x|_{\eta=0}$ and $\tau_y|_{\eta=0}$ increase on increasing K^2 . $\tau_x|_{\eta=0}$ increases on increasing M^2 . $\tau_y|_{\eta=0}$ increases when $K^2 = 2$ and it decreases when $K^2 = 3$ and 4 on increasing M^2 . This implies that, for $\omega > N^2$, rotation tends to enhance both the primary and secondary shear stress at the lower plate. Magnetic field tends to enhance primary shear stress at the lower plate whereas it has reverse effect on the secondary shear stress at the lower plate when $K^2 \geq 3$. It is observed from Table 7 that, for $\omega < N^2$, $\tau_x |_{\eta=1}$ decreases whereas $\tau_y|_{\eta=1}$ increases, attains a maximum and then increases on increasing K^2 . $\tau_x|_{\eta=1}$ and $\tau_y|_{\eta=1}$ decreases on increasing M^2 . This implies that, for $\omega < N^2$, rotation tends to reduce primary shear stress at the upper plate. Magnetic field tends to reduce both the primary and secondary shear stress at the upper plate.

Table 6 Shear stress at the lower plate due to primary and secondary flows when $m = 0.5$

and $\omega = 15(\omega > N^2)$

	$M^2 \downarrow K^2 \rightarrow$	2	3	
	10	1.1021	1.2890	1.4972
$\tau_{x} _{\eta=0}$	15	1.9593	2.1080	2.2695
	20	2.6682	2.7893	2.9189
	10	1.7643	2.1373	2.4744
$\tau_{y} _{\eta=0}$	15	1.8020	2.1065	2.3867
	20	1.8372	2.0972	2.3405

Table 7 Shear stress at the upper plate due to primary and secondary flows when $m = 0.5$

and $\omega = 5(\omega < N^2)$

	$M^2 \downarrow K^2 \rightarrow$	2	3	4
	10	0.2588	0.1618	0.0718
$\tau_{x} _{\eta=1}$	15	0.1365	0.0820	0.0316
	20	0.0740	0.0414	0.0112
	10	0.3672	0.3894	0.3859
$-\tau_{y} _{\eta=1}$	15	0.2431	0.2509	0.2464
	20	0.1662	0.1683	0.1640

It is found from Table 8 that, for $\omega > N^2$, $\tau_x |_{\eta=1}$ decreases whereas $\tau_y|_{\eta=1}$ increases on increasing K^2 . $\tau_x|_{\eta=1}$ and $\tau_y|_{\eta=1}$ decrease on increasing M^2 .

This implies that, for $\omega > N^2$, rotation tends to reduce primary shear stress at the upper plate whereas it has reverse effect on secondary shear stress at the upper plate. Magnetic field has tendency to reduce both the primary and secondary shear stress at the upper plate.

Table 8 Shear stress at the upper plate due to primary and secondary flows when $m = 0.5$

and $\omega = 15(\omega > N^2)$	
---------------------------------	--

6. CONCLUSION

The Present investigation deals with the theoretical study of unsteady MHD Couette flow of class-II in a rotating system. The significant results are summarized below:

- 1. Hall current tends to retard primary flow whereas it has reverse effect on secondary flow.
- 2. Rotation tends to accelerate both the primary and secondary flows.
- 3. Magnetic field tends to accelerate primary flow whereas it has reverse effect on secondary flow.
- 4. Oscillations tend to retard primary flow when $\omega \neq N^2$ and secondary flow when $\omega > N^2$ whereas it has reverse effect on secondary flow when $\omega < N^2$.
- 5. For $\omega < N^2$, Hall current and oscillations tend to reduce primary shear stress at the lower plate whereas it have reverse effect on secondary shear stress at the lower plate.
- 6. For $\omega > N^2$, Hall current tends to enhance secondary shear stress at the lower plate whereas it has reverse effect on the primary shear stress at the lower plate when $\omega \le 15$. Oscillations tend to reduce primary shear stress at the lower plate when $m \leq 1.0$ and secondary shear stress at the lower plate behaves in oscillatory manner with respect to oscillations. There exists flow separation at the lower plate in primary flow direction on increasing either m or ω .
- 7. For $\omega < N^2$, Hall current tends to enhance secondary shear stress at the upper plate whereas oscillations have tendency to enhance both the primary and secondary shear stress at the upper plate.
- 8. For $\omega > N^2$, Hall current tends to enhance both the primary and secondary shear stress at the upper plate and oscillations tends to reduce secondary shear stress at the upper plate while it tends to enhance primary shear stress at the upper plate when $m \leq 1.0$.
- 9. For $\omega < N^2$, rotation tends to enhance both the primary and secondary shear stress at the lower plate. Magnetic field tends to enhance primary shear stress at the lower plate when $K^2 \geq 3$. Secondary shear stress at the lower plate behaves in oscillatory manner with respect to magnetic field.
- 10. For $\omega > N^2$, rotation tends to enhance both the primary and secondary shear stress at the lower plate. Magnetic field tends to enhance primary shear stress at the lower plate whereas it has reverse effect on the secondary shear stress at the lower plate when $K^2 \geq 3$.
- 11. For $\omega < N^2$, rotation tends to reduce primary shear stress at the upper plate. Magnetic field tends to reduce both the primary and secondary shear stress at the upper plate.
- 12. For $\omega > N^2$, rotation tends to reduce primary shear stress at the upper plate whereas it has

reverse effect on secondary shear stress at the upper plate. Magnetic field has tendency to reduce both the primary and secondary shear stress at the upper plate.

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